

Attorney Docket No. 82649  
Customer No. 23523

**GPS ANTENNA FOR SUBMARINE TOWED BUOY**

TO ALL WHOM IT MAY CONCERN:

BE IT KNOWN THAT DAVID F. RIVERA, employee of the United States Government, citizen of the United States of America, and resident of Westerly, County of Washington, State of Rhode Island has invented certain new and useful improvements entitled as set forth above of which the following is a specification:

MICHAEL P. STANLEY  
Reg. No. 47108  
Naval Undersea Warfare Center  
Division, Newport  
Newport, RI 02841-1248  
TEL: 401-832-4736  
FAX: 401-832-1231

2

3 **GPS ANTENNA FOR SUBMARINE TOWED BUOY**

4

5 **STATEMENT OF GOVERNMENT INTEREST**

6 The invention described herein may be manufactured and used  
7 by or for the Government of the United States of America for  
8 governmental purposes without the payment of any royalties  
9 thereon or therefor.

10

11 **BACKGROUND OF THE INVENTION**

12 **(1) Field of the Invention**

13 The present invention relates to antennas and more  
14 particularly to a global positioning system (GPS) antenna.

15 **(2) Description of the Prior Art**

16 In the field of GPS technology, GPS receivers are used to  
17 determine the geographic location of the receiver by receiving  
18 microwave radio signals from a group of earth-orbiting GPS  
19 satellites. The geographic location of the receiver may be  
20 computed by calculating its distance from each satellite as the  
21 result of determining how long the signals take to travel from  
22 the satellite to the receiver. Typically, a flat GPS antenna  
23 element is utilized by GPS receivers to receive the signals  
24 transmitted. In order for the GPS receiver to compute its

1 geographic location, the antenna element of the receiver must be  
2 oriented to receive an acceptable level of the signals.  
3 Optimally, the flattened surface of the GPS antenna element is  
4 righted against the force of gravity such that a maximum surface  
5 area of the antenna faces the satellites.

6 Present submarine communications with battlegroups or  
7 satellites utilize surface antennas for a variety of  
8 requirements including global positioning and communications.  
9 The use of surface antennas typically interferes with the covert  
10 operation of the submarine. For example, submarines obtaining  
11 position fixes using GPS must raise a mast containing an antenna  
12 which is oriented to receive the signals from the GPS  
13 satellites. The problem is that raising a mast renders the  
14 submarine vulnerable to either visual or radar detection,  
15 especially if the mast is raised in coastal or littoral areas.

16 Additionally, antennas used on the ocean surface are  
17 subjected to dynamic forces that act to cause the antenna to  
18 pitch, yaw and sometimes roll with the vessel under varying sea  
19 states. These antenna movements can easily re-orientate the  
20 receiving element of the antenna resulting in reception  
21 interruption. Varying sea states also cause a detuning effect  
22 that result in degradation of the patch elements of conventional  
23 GPS antennas. To minimize the effects of varying sea states,

1 the submarine must operate in a station keeping status or must  
2 constantly adjust course headings.

3 One method of mitigating reception interruption of the  
4 antenna is to orient the flattened surface of the antenna to  
5 right itself or face "up" toward the sky irrespective of the  
6 movement of its supporting structure. In Ham (U.S. Patent No.  
7 6,292,147), an apparatus for maintaining a GPS antenna element  
8 at a predetermined orientation is disclosed. The apparatus  
9 includes a holder configured to support a GPS antenna element in  
10 which the holder includes a rectangular frame as a receiving  
11 portion of the dielectric substrate of antenna. The rectangular  
12 holder pivots on an axis in relation to gravity to the  
13 predetermined orientation even when the base structure to which  
14 the holder is coupled changes its orientation. While the  
15 disclosed reference allows a righting motion to the antenna  
16 element, the movement of the righting motion is limited to  
17 rotation around the axis of the pivot in which the rotation  
18 provides only one degree of freedom.

19 It is well known in the use of gyroscopes and in the use of  
20 compasses on ships, that a gimbal provides at least two degrees  
21 of freedom for either attached device by allowing a pivoting  
22 action on the axes of the gimbal in which the axes are rotatable  
23 at angles to each other. For example, the pivoting and rotating  
24 action of a gimbal used on a ship compensates for the roll and

1 the yaw of the ship as well as the pitch of the ship thereby  
2 maintaining an accurate heading of a compass set in the gimbal.

3 As such, an improvement to the technology of GPS antennas  
4 would be to incorporate the degrees of freedom of a gimbal with  
5 a conformable GPS antenna in a manner that is suitable for use  
6 on a vessel or towed array as well as for use in any other  
7 situation that can require more than one degree of freedom in  
8 which the degree of freedom is needed to maintain the righting  
9 or facing up element of the antenna receiver. Such an  
10 improvement along with any other suitable improvements to the  
11 structure of the GPS antenna could act to minimize the reception  
12 interruptions and the detuning effects caused by varying sea  
13 states.

14

#### 15 **SUMMARY OF THE INVENTION**

16 Accordingly, it is a general purpose and primary object of  
17 the present invention to provide an apparatus with a Global  
18 Positioning System (GPS) antenna that can obtain geographic  
19 positioning data with minimal interruption when operating in  
20 varying sea states.

21 It is a further object of the present invention to provide  
22 an apparatus with an antenna that can transmit and receive  
23 signal communications with minimal interruption when operating  
24 in varying sea states.

1       It is a still further object of the present invention to  
2 provide an apparatus with antenna that can be towed by a  
3 submarine.

4       It is a still further object of the present invention to  
5 provide an apparatus with antenna in which the construction is  
6 simple and economical.

7       It is a still further object of the present invention to  
8 provide an antenna capable of transmission at high frequencies  
9 with minimal degradation.

10       It is a still further object of the present invention to  
11 provide an antenna in which the construction is simple and  
12 economical.

13       To attain the objects described, there is provided an  
14 apparatus with a GPS antenna in which the antenna maintains a  
15 receiving area that faces toward the sky or ocean surface. The  
16 antenna is a hollowed frustum having a closed end at its  
17 decreased diameter and an integral base ring surrounding an open  
18 end at an increased diameter of the frustum. The antenna  
19 includes a feed stem at the closed end extending as an internal  
20 rod in the interior of the frustum. The opposite end of the  
21 internal rod connects to a receiver plate in which the receiver  
22 plate extends from the base ring toward and beyond a  
23 longitudinal axis of the frustum.

1        For use in vessel operations or other applications that  
2        require the receiver plate to face the sky or the ocean surface,  
3        the antenna is supported by a gimbal. The gimbal is attachable  
4        to the interior of a watertight container suitable for towing  
5        horizontally on the ocean surface.

6        During operations, the pivoting of the antenna at the open  
7        end in relation to the lower center-of-gravity of the frustum  
8        shape of the antenna allows an enhanced swinging arc in relation  
9        to the attached gimbal in that the body of the frustum moves by  
10       gravity toward the axes of the gimbal. As such, the antenna  
11       provides the righting or facing up of the open end of the  
12       frustum and a facing up of the flattened surface of the attached  
13       receiver plate thereby permitting enhanced reception by the  
14       antenna. Furthermore, the antenna itself and not a holder of  
15       the antenna provides the righting or facing up motion thereby  
16       allowing a reduction in the amount of parts and a simplicity in  
17       design.

18       During actuation of the antenna, the feed stem is  
19       conductive to an energized feed source. Radio-frequency energy  
20       from the feed stem continues to the frustum with the energy  
21       disbursing as a current distribution along the interior surface  
22       of the frustum. The radio-frequency energy from the feed stem  
23       also continues onto the receiver plate with the result of a  
24       current distribution across the receiver plate. The differences

1 in phase and amplitude from the radiating surface of the  
2 frustum, and the receiver plate contributes to a hemispherical  
3 radiation pattern in the far field.

4 The hemispherical radiation pattern is advantageous because  
5 when the antenna is placed on the ocean surface, the radiation  
6 pattern in the air space above the ocean surface does not  
7 contain nulls. As such, the radiation pattern in the air space  
8 permits full directionalized reception from GPS satellites or  
9 other signal emitting sources.

10 Furthermore, the antenna of the present invention reduces  
11 the degradation and associated problems with detuning occurring  
12 during various sea states. Specifically, the impedance matching  
13 of the frustum shape and the components of the antenna control  
14 the impedance influence of the detuning. Also, the structure of  
15 the curved frustum shape removes the edges of a typical patch  
16 antenna in which the edges of the typical patch antenna are  
17 subject to degradation from detuning.

18 The above and other features of the invention, including  
19 various and novel details of construction and combinations of  
20 parts will now be more particularly described with reference to  
21 the accompanying drawings and pointed out in the claims. It  
22 will be understood that the particular devices embodying the  
23 invention are shown by way of illustration only and not as the  
24 limitations of the invention. The principles and features of



1 this invention may be employed in various and numerous  
2 embodiments without departing from the scope of the invention.

#### 3 4 **BRIEF DESCRIPTION OF THE DRAWINGS**

5 A more complete understanding of the invention and many of  
6 the attendant advantages thereto will be readily appreciated as  
7 the same becomes better understood by reference to the following  
8 detailed description when considered in conjunction with the  
9 accompanying drawings wherein:

10 FIG. 1 is a side view of the antenna of the present  
11 invention;

12 FIG. 2 is a plan view of the antenna of the present  
13 invention with the view taken from reference line 2-2 of FIG. 1;

14 FIG. 3 is an alternate plan view of the antenna of the  
15 present invention with the view taken from reference line 3-3 of  
16 FIG. 1;

17 FIG. 4 is a side view of the antenna of the present  
18 invention with the antenna mounted on a gimbal positioned in an  
19 antenna housing;

20 FIG. 5 is a cross-sectional view of the antenna housing  
21 attached to a tow body with the view taken from reference line  
22 5-5 of FIG.4; and

23 FIG. 6 is a three dimensional view of a radiation pattern  
24 formed by the antenna of the present invention.

## DESCRIPTION OF THE PREFERRED EMBODIMENT

Referring now to the drawings wherein like numerals refer to like elements throughout the several views, one sees that FIG. 1 depicts the antenna 10 of the present invention. The antenna 10 is preferably cast with a rigid thickness from phosphor bronze or beryllium copper with electrically conductive components attached or also cast as part of the antenna. Other commonly acquired materials resistant to corrosion in a sea environment or materials known to those skilled in the art may be used in forming the antenna 10.

The simplified structure of the antenna 10 comprises a hollowed frustum 12 having an open end 14 and a closed end 16 with a distance between the closed end and the open end being approximately  $\lambda/9$ , wherein  $\lambda$  is the free-space wavelength measured in meters. For GPS use, the free-space wavelength equals the center frequency of operation, [the square root of the multiplication of the GPS frequencies (1227 MHz, 1575 MHz)] divided by the speed of light. The sizing of the diameter of the frustum 12 as well as the sizing of the other components of the antenna 10 is based on the free-space wavelength thereby allowing the antenna to be sized at a substantial bandwidth for alternate functions such as receiving and transmitting signals from IRIDIUM satellites (1625 MHz).

1        For the open end 14 of the frustum 12 shown in FIG. 2, the  
2        open end has a diameter "A" of  $2\lambda/5$ . An integral base ring 18  
3        projects from the open end 14 parallel to a longitudinal axis 20  
4        of the antenna 10 in which the longitudinal axis is preferably  
5        perpendicular to the open end 14 and the closed end 16. The  
6        base ring 18 includes a notch 22 to position a receiver plate 24  
7        flush with the projection of the open end 14. The receiver  
8        plate 24 extends from the notch 22 to and beyond the  
9        longitudinal axis 20. The receiver plate 24 is generally  
10       rectangular in shape from the flush with the notch 22 with the  
11       rectangular shape having a nominal length "B" of  $\lambda/3$  and a width  
12       "C" that is approximately ten percent less than the length "B".

13       For the closed end 16 of the frustum shown in FIG. 3, the  
14       closed end 16 has a diameter of  $\lambda/5$ . The closed end 16 includes  
15       a feed stem 30 shielded by an extension 31 of the frustum 12.  
16       The feed stem 30 extends as an internal rod 32 in the cavity of  
17       the antenna 10. See FIG. 1. For an optimum impedance match and  
18       bandwidth to the antenna structure described above, the diameter  
19       of the rod 32 is  $\lambda/30$  with a length of  $\lambda/10$  and a contact point  
20       for the receiver plate 24 at  $\lambda/11$  from the plate edge 34. The  
21       depth of the cavity (noted above as the distance between the  
22       open end 14 and the closed end 16), the size (the length "B" and  
23       the width "C") of the receiver plate 24 and the size of the rod  
24       32 determine the impedance at the feed stem 30, the radiation

1 pattern 36 of the antenna 10 and the bandwidth of the antenna  
2 10.

3 Referring again to FIG. 2, the base ring 18 includes  
4 attachment points 40, 42 in which the points allow the insertion  
5 of a swivel axis or any other mechanical attachment to a gimbal  
6 50, described below. As shown in FIG. 4 for the use of the  
7 antenna 10 in submarine operations, the antenna 10 is supported  
8 by the gimbal 50 attached to the interior of the watertight  
9 container 52. The watertight container 52 is electrically  
10 transparent polyethylene and is attachable to a tow body  
11 54 (shown in FIG. 5) which can be towed by a submarine or other  
12 vessel.

13 The pivoting at the attachment points 40, 42 of the antenna  
14 10 in relation to the lower center-of-gravity of the frustum  
15 shape of the antenna allows an enhanced swinging arc by gravity  
16 (54) on the axis of the attachment points 40, 42 in relation to  
17 the attached gimbal 50. The gimbal 50 in turn has a swinging  
18 arc (56) on its own attachment points 58, 60; thereby providing  
19 a righting movement for the antenna 10 on at least two axes. As  
20 such, the antenna 10 provides the righting or facing up of the  
21 open end 14 of the frustum and a facing up of a flattened  
22 surface of the receiver plate 24 toward overhead satellites  
23 thereby permitting enhanced reception by the antenna. The  
24 antenna 10 is further unique in that the antenna itself and not

1 a holder of the antenna provides the righting or facing up  
2 motion thereby allowing a reduction in moving parts and a  
3 simplicity in design.

4 During actuation of the antenna 10, the feed stem 30 is  
5 conductive to an energized feed source (not shown). Radio-  
6 frequency energy from the feed stem 30 continues onto the  
7 frustum 12 with the energy disbursing as a current distribution  
8 along the interior surface of the frustum. The energy from the  
9 feed stem 30 also continues to the receiver plate 24 by way of  
10 the rod 32 with the result of a current distribution across the  
11 receiver plate. The distribution of current amplitude and phase  
12 from the surface of the frustum 12 and the receiver plate 24  
13 contributes to a hemispherical radiation or beam pattern 36,  
14 shown in FIG. 6. The hemispherical radiation pattern 36 is  
15 advantageous because when the antenna 10 is placed on the ocean  
16 surface, the radiation pattern in the air space above the ocean  
17 surface (shown by the area 76 above the "x" and "y" axis) does  
18 not contain nulls. As such, the radiation pattern in the air  
19 space permits full directionalized reception from satellites.

20 Furthermore, the antenna 10 reduces the degradation and  
21 associated problems with detuning occurring during with vary sea  
22 states. Specifically, the impedance matching of the frustum 12,  
23 the feed stem 30 and the rod 32 control the impedance influence  
24 of the detuning. Also, the structure of the curved frustum 12

1 removes the edges of a typical patch antenna in which the edges  
2 of the typical patch antenna are subject to detuning and quicker  
3 degradation.

4 An additional feature of the present invention is that the  
5 structural ratio (identified by the wavelength dimensioning  
6 above) of the various components of the antenna 10 allows the  
7 hemispherical radiation pattern 36 while maintaining the  
8 compactness of the antenna 10. The compactness of the antenna  
9 10 is advantageous for many reasons including detection  
10 minimalization and reduced drag of the enclosing towing body.  
11 In relation to conventional GPS antennas, the compactness of the  
12 antenna 10 with its frustum 12 and receiver plate 24 does not  
13 require a large ground plane in order to generate the  
14 hemispherical radiation pattern 36.

15 In defining the compactness feature, the outer physical  
16 boundary of the antenna 10 is based on the size and placement of  
17 the open end 14 and the closed end 16 of the frustum 12. For  
18 example, the diameters of the open end 14 and the closed end 16  
19 are  $2\lambda/5$  and  $\lambda/5$  respectively with a distance of  $\lambda/9$  between the  
20 open end and the closed end. Any remaining structure of the  
21 antenna 10 would be within a circumferential boundary created by  
22 the above dimensions.

23 Furthermore, the all-metallic structure of the antenna 10  
24 does not require a ceramic dielectric substrate yet allows

1 transmission and reception at a large instantaneous operating  
2 bandwidth as exemplified by the antenna use with IRIDIUM and  
3 global positioning signals described above.

4       Thus by the present invention its objects and advantages  
5 are realized and although preferred embodiments have been  
6 disclosed and described in detail herein, its scope should be  
7 determined by that of the appended claims.